

# Jets from variable sources

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VLT-FORS2

(1999)

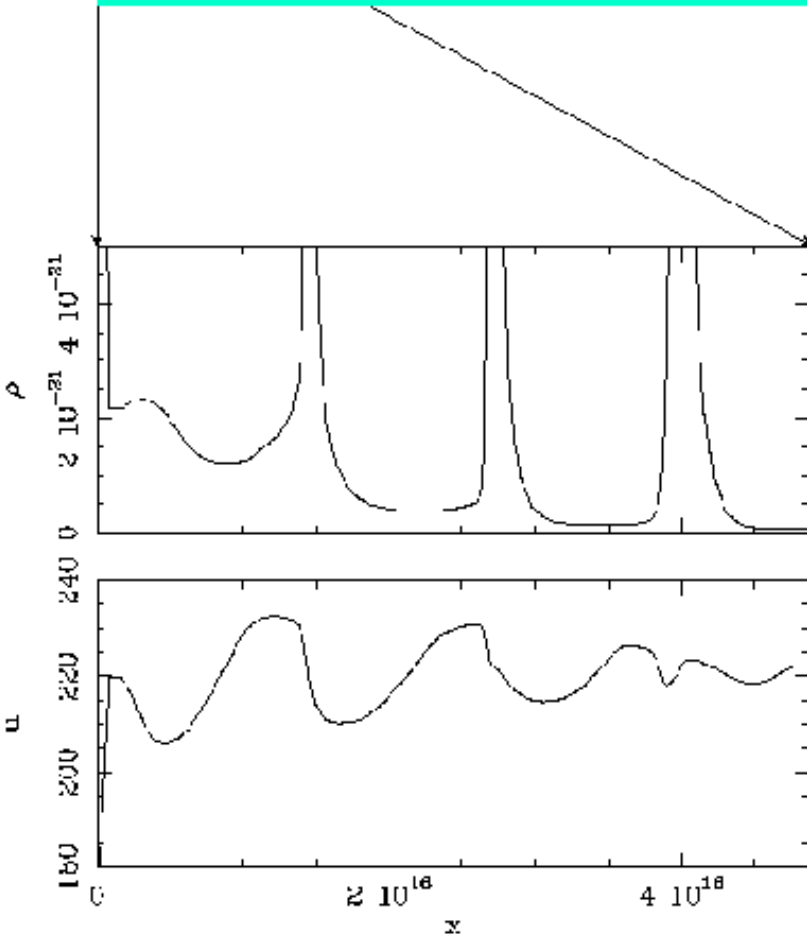
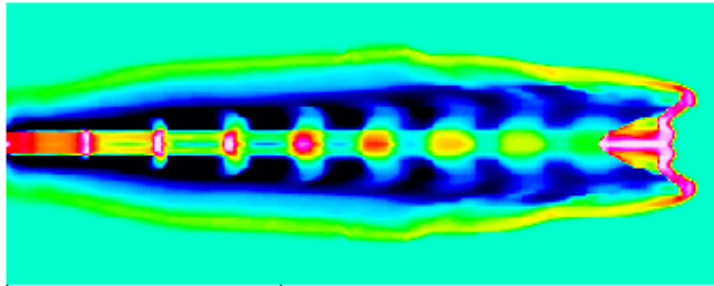
## Ejection velocity variability

→ Leads to the formation of “internal working surfaces”

## Density variability

☐ Leads only to the formation of weak shocks

## Jet with variable ejection velocity

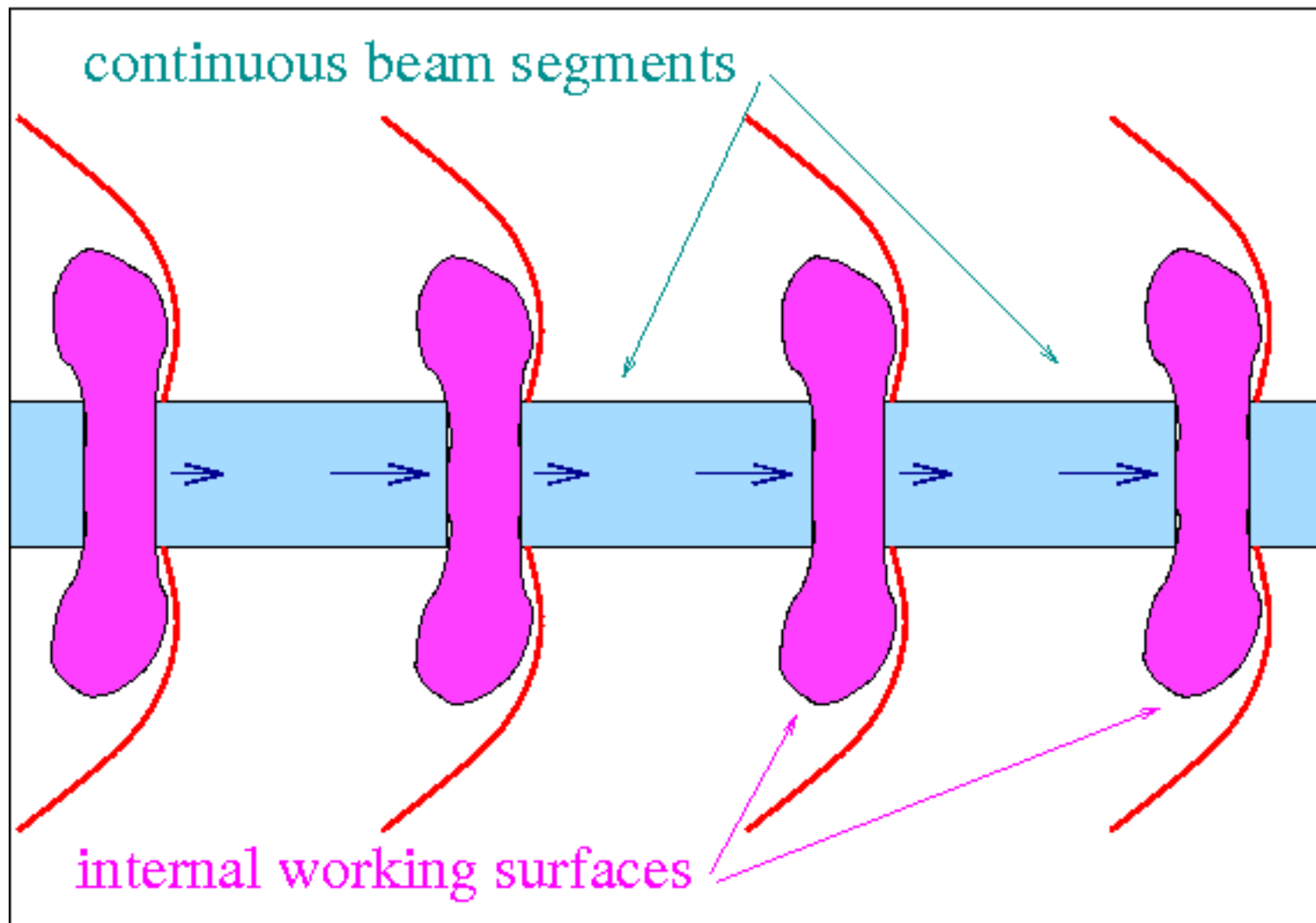


$$v_j = v_0 + \Delta v \sin(2\pi t/\tau)$$

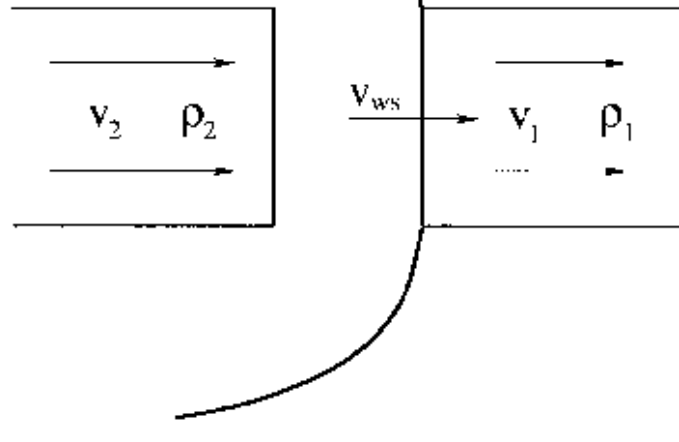
$$v_0 = 200 \text{ km/s}, \quad \Delta v = 40 \text{ km/s}, \quad \tau = 20 \text{ yr}$$

$$n_j = 1000 \text{ cm}^{-3}, \quad T_j = 1000 \text{ K}, \quad r_j = 2 \times 10^{15} \text{ cm}$$

$$n_{\text{env}} = 100 \text{ cm}^{-3}, \quad T_{\text{env}} = 1000 \text{ K}$$



## Free-streaming continuous beam segments



$$[?] \quad u(x, t) = u_0(\tau) = \frac{x}{t - \tau}$$

$$\sigma(x) = \sigma_0 \left( \frac{x_0}{x + x_0} \right)^P \quad \text{pos-dep cross section}$$

$$[?] \quad \rho(x, t) = \rho_0(\tau) \left( \frac{x_0}{x + x_0} \right)^P \left[ 1 - (t - \tau) \frac{d \ln u_0}{d \tau} \right]^{-1}$$

## E.O.M. for internal working surface

Ram pressure balance:

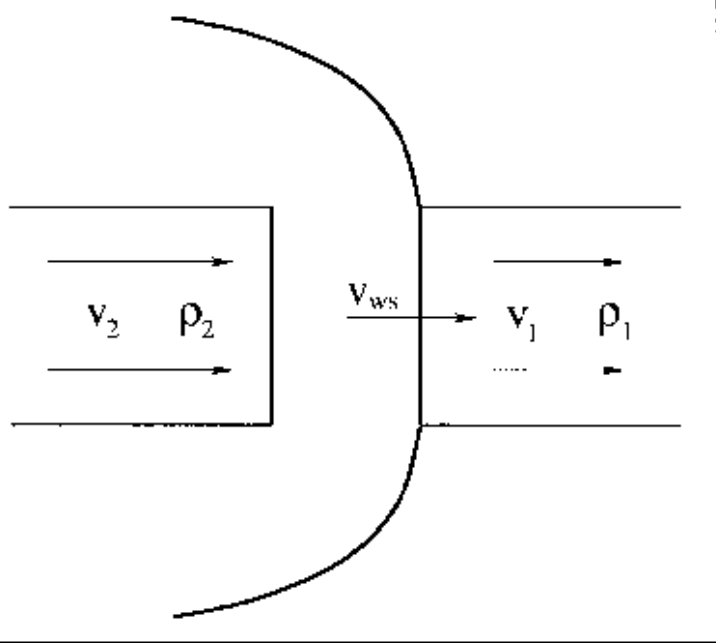
$$v_s = \frac{dx_s}{dt} = \frac{u_1 + \beta u_2}{1 + \beta}, \quad \beta = \sqrt{\frac{\rho_2}{\rho_1}}$$

Centre of mass formalism:

$$v_{ws} = \frac{1}{m} \int_{\tau_1}^{\tau_2} \dot{m}(\tau) v_0(\tau) d\tau.$$

$$x_{ws} = (t - \tau_2) v_{ws} + \frac{1}{m} \int_{\tau_1}^{\tau_2} \dot{m}(\tau) v_0(\tau) (\tau_2 - \tau) d\tau,$$

$$m = \int_{\tau_1}^{\tau_2} \dot{m}(\tau) d\tau.$$



## Ram-pressure balance eom:

→ Asymptotic solution for large distances from the source

(Kofman, Raga 1998)

## Centre of mass eom:

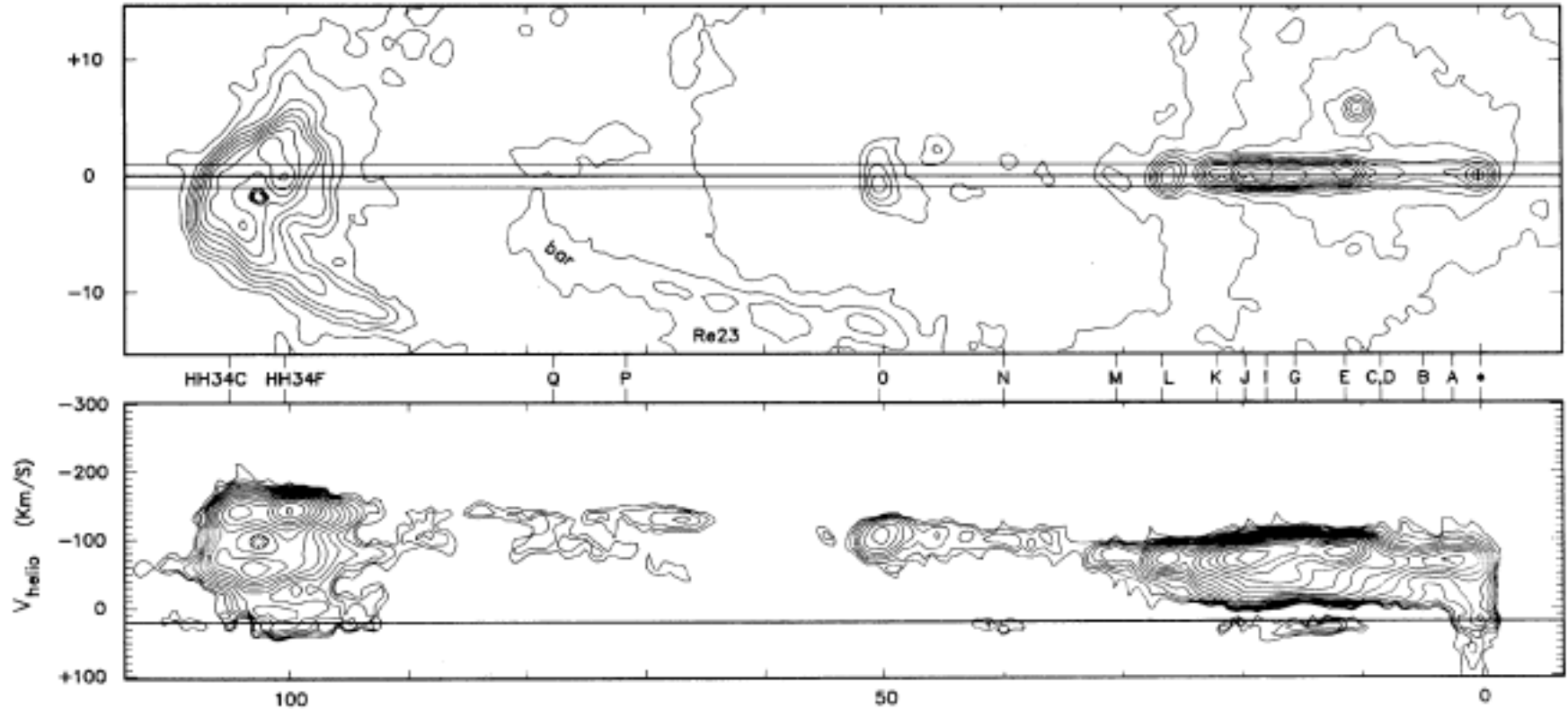
→ All kinds of analytic solutions

(Canto et al. 2000)

**2 – 3D numerical simulations produce results which lie between these two analytic approximations**

## Observational evidence for ejection time-variability:

1. “acceleration” observed along some HH jets



HH 34: Heathcote and Reipurth (1992)

Initial jet velocity  $\approx$  200 km/s

Initial jet sound speed  $\approx$  10 km/s

Final jet speed  $\approx$  300 km/s

$$v^2/2 + c_s^2/(\gamma-1) = \text{const.}$$

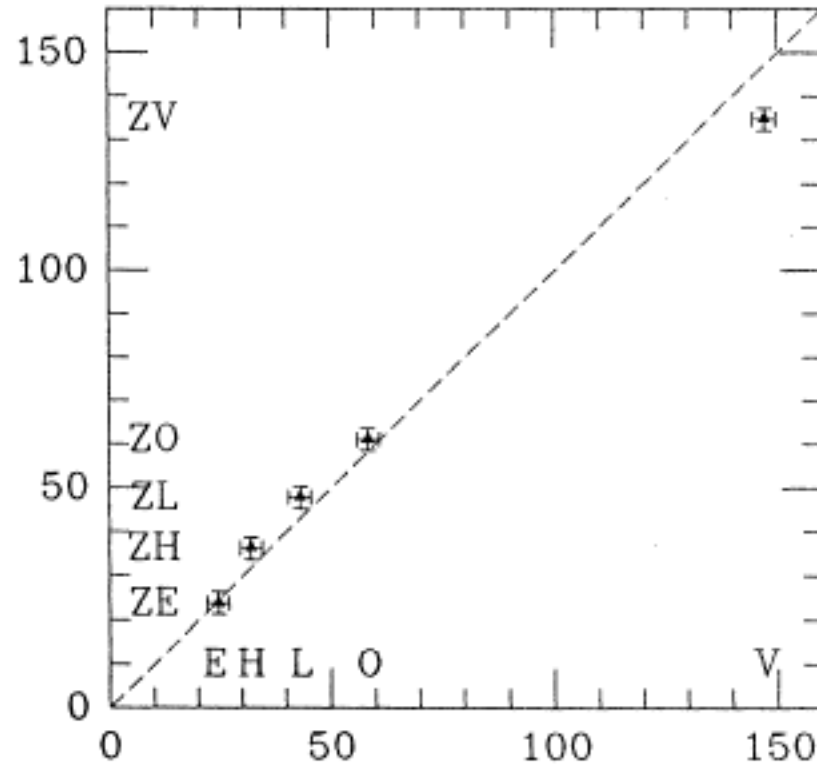
$$M_0 = v_0/c_{s,0} \approx 20$$

→ Impossible to produce the observed acceleration in a steady jet

$\approx$  ejection velocity varies with time

## Observational evidence for ejection time-variability:

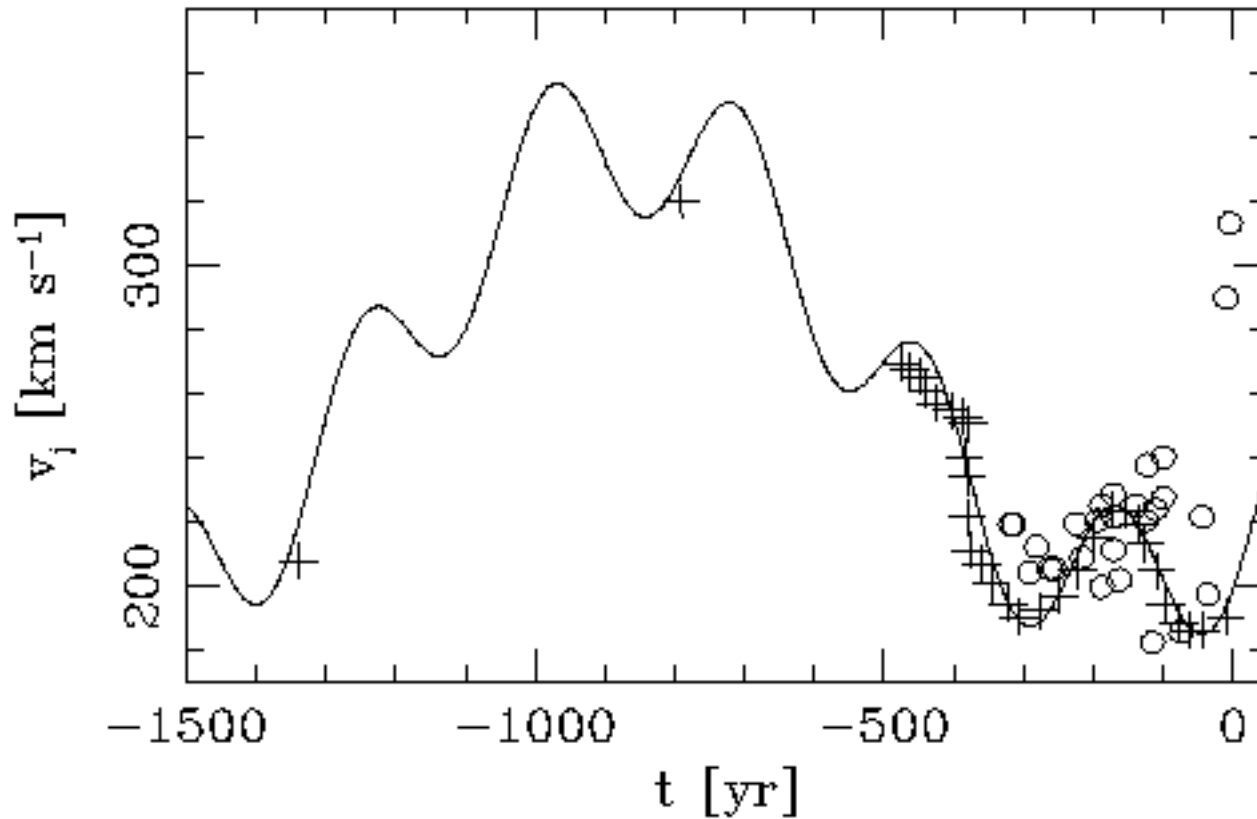
2. “point symmetry” between knots in jet and counterjet



HH 111: Gredel and Reipurth (1994)

10/09/09 (HH 34 ? Panoglou et al. 2009)

Free-streaming flow can be used to reconstruct ejection velocity history



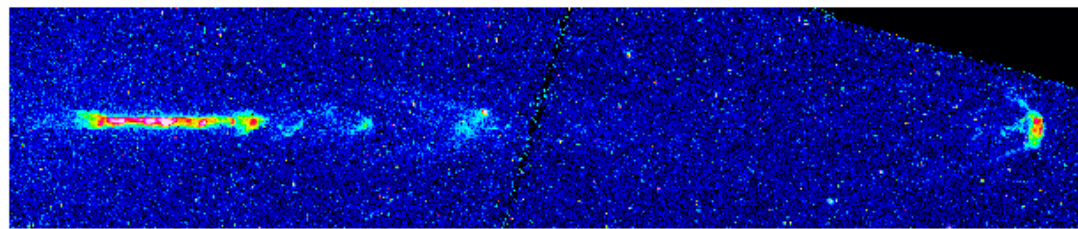
HH 34: Raga et al. (2002)

$$u(x, t) = u_0(\tau) = \frac{x}{t - \tau}$$

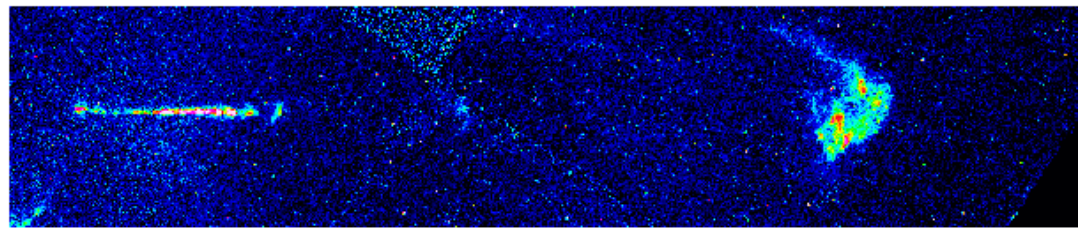
# Reconstructed ejection velocity variability

→ Fed into a numerical simulation

HF 111

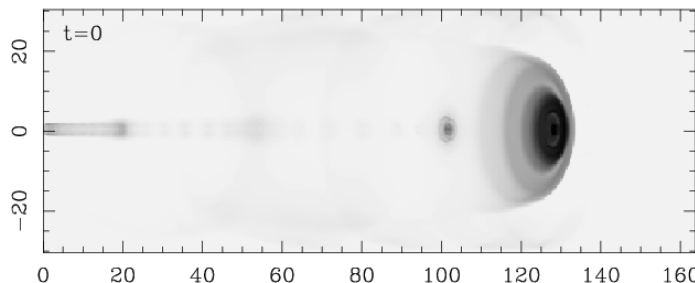
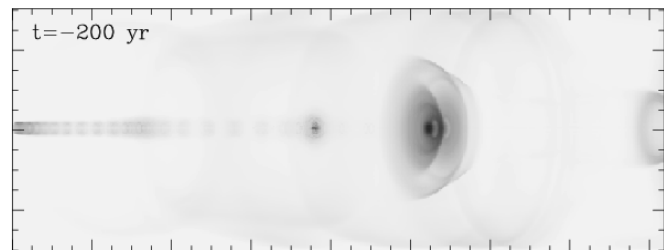
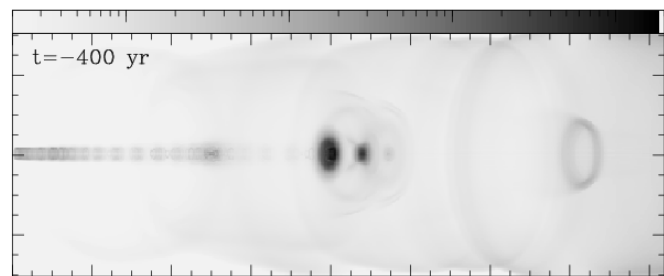


HF 34

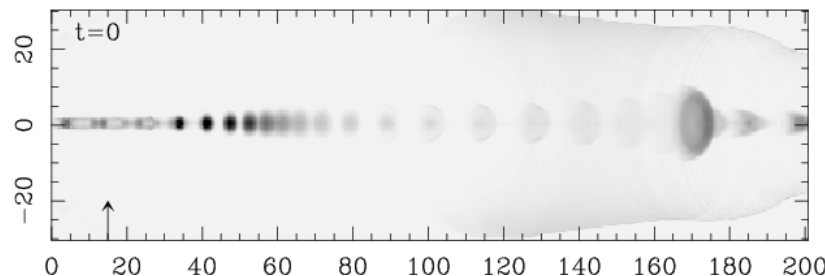
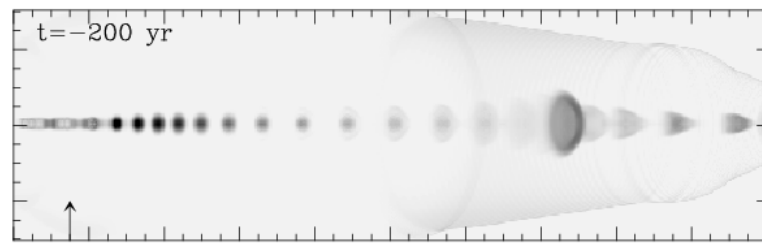
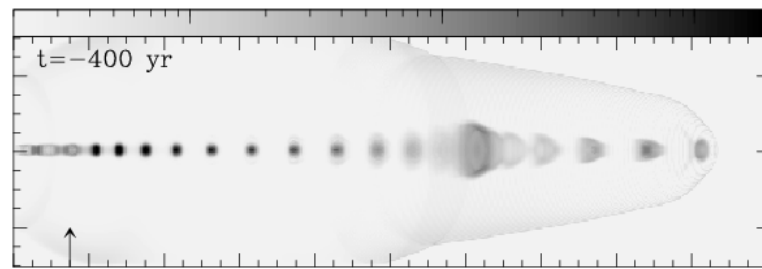


30''

$10^{-6}$     $10^{-5}$     $10^{-4}$     $10^{-3}$



$10^{-5}$     $10^{-4}$     $10^{-3}$



Raga et al.  
(2002)

# Ejection velocity variability + precession

**3D:**  $1024 \times 512 \times 512$

→ size of domain along axis:  $4 \times 10^{17}$  cm

→ jet radius:  $r_j = 7 \times 10^{15}$  cm

→ jet radius resolved with 18 grid points

→  $v_j = 200 \text{ km s}^{-1}$ ,  $n_j = 1000 \text{ cm}^{-3}$ ,  $T_j = 1000 \text{ K}$

→  $n_{env} = 100 \text{ cm}^{-3}$ ,  $T_{env} = 100 \text{ K}$

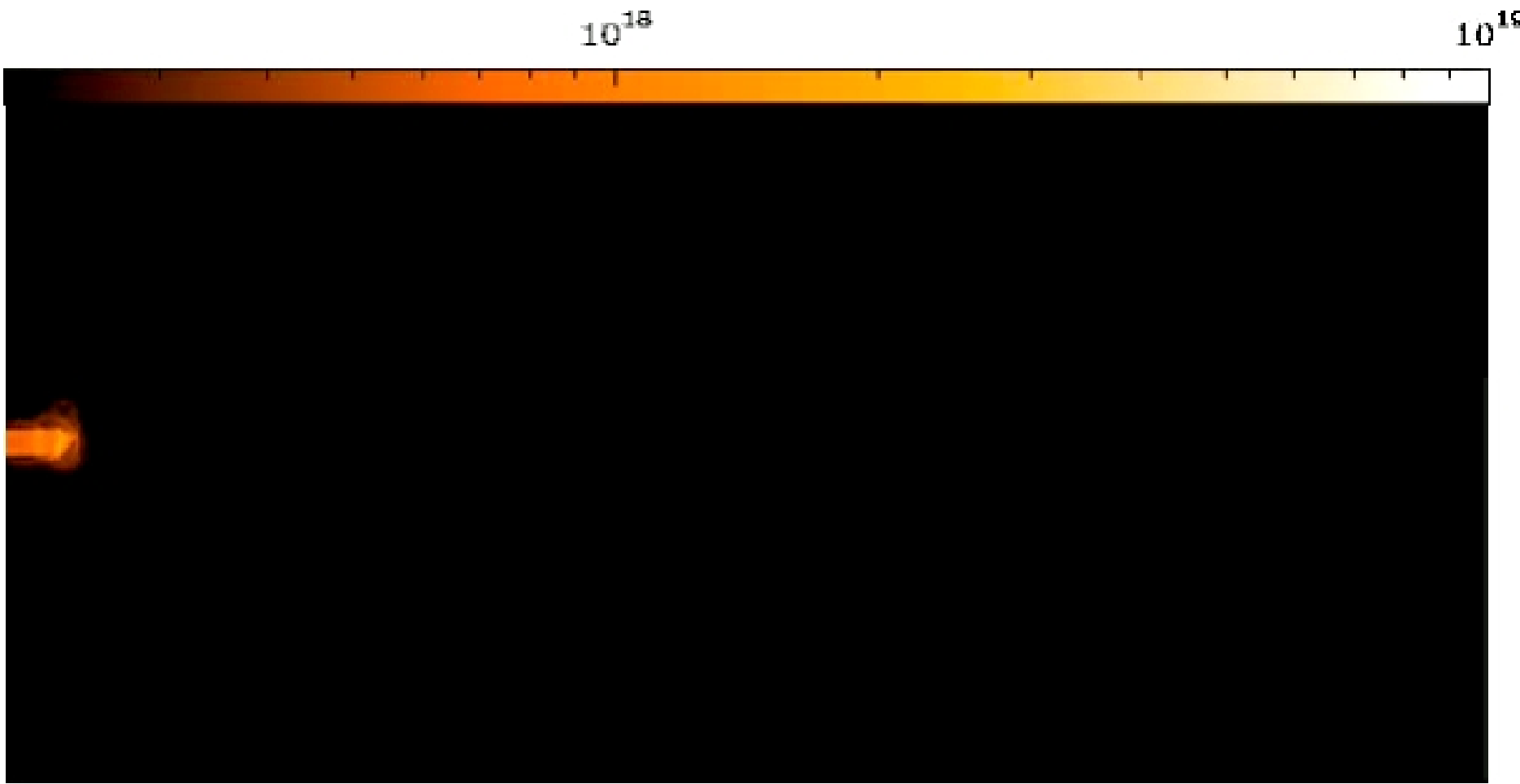
→ **two sinusoidal modes:**

$$\Delta v = 20 \text{ km s}^{-1}, \tau = 30 \text{ yr}$$

$$\Delta v = 100 \text{ km s}^{-1}, \tau = 230 \text{ yr}$$

→ **precession:**  $\tau_{prec} = 1000 \text{ yr}$ , half-angle:  $5^\circ$

# “modern” 3D numerical simulation



t=50 yr

## Summary:

### Evidence that HH jets are the result of a time-dependent ejection

- Multiple “heads” (e. g. in giant jets)
- “point symmetric” chains of knots
- “accelerations”

### Theoretical work

- Analytic models (ram pressure balance, centre of mass)
- Numerical simulations (1,2,3D; multiple mode, precession, sidewind, orbital motion, externally photoionized jets, stratified medium, interaction with clumps, clumpy jet)

# Combining theory+observations

1. Observed kinematics → reconstruction of ejection history through analytic model

2. Numerical simulation → prediction of observed emission

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